

**TITLE**

Arrangement and method for producing a three-dimensional product.

**TECHNICAL FIELD**

- 5 The invention relates to an arrangement and a method for production of a three-dimensional product by successive fusing together of selected parts of powder layers applied to a work table.

**BACKGROUND ART**

- 10 An arrangement for producing a three-dimensional product by successive fusing together of selected parts of powder layers applied to a work table is previously known from, for example, US 4863538. The arrangement comprises a work table on which said three-dimensional product is to be built up, a powder dispenser which is arranged so as to distribute a thin layer of powder on the work table for forming a powder bed, a radiation gun for delivering energy to the powder, fusing together of the powder then taking place, means for guiding the beam emitted by the radiation gun over said powder bed for forming a cross section of said three-dimensional product by fusing together parts of said powder bed, and a control computer in which
- 15 information about successive cross sections of the three-dimensional product is stored. The three-dimensional product is built up by fusing together selected parts of successive powder layers applied. The control computer is intended to control deflection means for the beam generated by the radiation gun over the powder bed according to an operating scheme which
- 20 reproduces a predetermined pattern. When the operating scheme has fused together a desired area of a powder layer, a cross section of said three-dimensional product has been formed. A three-dimensional product is formed by successive fusing together of successively formed cross sections from powder layers applied successively by the powder dispenser.
- 25
- 30 An arrangement for producing a three-dimensional product where measurement of the surface structure and the surface temperature of the

three-dimensional body produced is permitted during the manufacturing procedure is known from SE 0001557-8. By using the arrangement described therein, increased correspondence of the shape of the three-dimensional bodies produced in relation to the intended shape is made possible.

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In order to meet industrial requirements in the manufacture of three-dimensional models, it is important to achieve a sufficiently high manufacturing rate. This is especially the case if an arrangement as described above is to be used for series production. Attempts have been 10 made using increased sweep speed with a corresponding increase in power of the beam. In the process for manufacturing the three-dimensional products, however, it has been found that surface stresses in the manufactured product give rise to shape deviations and also internal stresses in the product which can give rise to the initiation of crack formation.

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#### BRIEF DESCRIPTION OF THE INVENTION

One object of the invention is to provide a method for production of three-dimensional bodies where the manufacturing process for the three-dimensional body can be rendered more effective for reduction of 20 manufacturing times. This object is achieved by an arrangement according to the characterizing part of patent claim 1.

By virtue of, in a method for production of three-dimensional bodies, where a 25 selected area corresponds to a cross section of the finished product, using two or more fusion zones which propagate simultaneously through the selected area when formation of a cross section of the three-dimensional body takes place, rendering manufacture more effective is made possible.

In a first preferred method, said two or more fusion zones are created by a 30 radiation gun supplying energy to two geometrically separate focal points while time sampling takes place.

In another preferred embodiment, the focal points of the radiation gun propagate at said two fusion points at a speed which corresponds to the wave propagation speed of the fusion zone. By means of this procedure step, it is  
5 ensured that the risk of overheating is reduced, the end product then having a good surface finish.

In one embodiment of the invention, the wave propagation speed is estimated by measuring the wave propagation speed of the fusion zone from  
10 information provided by means for sensing the temperature distribution of a surface layer located in the powder bed.

In an alternative embodiment, the wave propagation speed is estimated by calculating an energy balance for an area comprising said focal points, said  
15 wave propagation speed being obtained from a model of a thermal conductivity equation set up for said area.

Both the proposed methods give a good idea of the wave propagation speed, it then being possible for a feedback system for maintaining correct  
20 propagation speed to be produced.

According to a preferred embodiment of the invention, an energy balance is calculated for each powder layer, it being determined in the calculation whether energy fed into the powder layer when said supply of energy from a  
25 radiation gun according to an operating scheme determined for the powder layer for fusing together that area of the powder layer selected according to said operating scheme takes place is sufficient to maintain a defined working temperature of the next layer, information being obtained which makes it possible to maintain a defined temperature. By maintaining a defined working  
30 temperature, that is to say a surface temperature within a given defined temperature range, during the production of all the layers, it is ensured that the occurrence of surface stresses which arise when cooling of the three-

dimensional body is too great is reduced. This in turn leads to the end product having a reduced occurrence of shape deviations and also a reduced occurrence of internal stresses in the end product.

- 5 The invention also relates to an arrangement for producing a three-dimensional product which has features corresponding to the method.

#### DESCRIPTION OF FIGURES

The invention will be described in greater detail below in connection with  
10 accompanying drawing figures, in which:

- fig. 1 shows a cross section of an arrangement according to the invention,
- 15 fig. 2 shows diagrammatically a cross section of a three-dimensional body formed by a number of powder layers and also a top powder layer,
- fig. 3 shows a schematic model for calculating energy balance,  
20
- fig. 4 shows another schematic model for calculating energy balance,
- fig. 5 shows the movement of focal points and fusion zones over a selected surface,  
25
- fig. 6 shows a selected area which is divided into a set of separate areas,
- fig. 7 shows a division of the separate areas into inner areas and edges,  
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- fig. 8 shows an area to be fused together, which has an inner area and an edge,
- 5 fig. 9 shows a set of different curve shapes with a one-dimensional interference term,
- 10 fig. 10 shows diagrammatically how the heat distribution appears in a body where the focal point with the diameter D of a radiation gun has heated the body, on the one hand in the presence of an interference term, on the other hand in the absence of an interference term,
- 15 fig. 11 shows an example of the movement of the focal point in relation to movement of the focal point along the main movement direction,
- 20 fig. 12 shows a set of different curve shapes with a two-dimensional interference term,
- 25 fig. 13 shows the movement pattern of a focal point according to a preferred embodiment of the invention,
- fig. 14 shows the positioning of the focal points and also a widened area within which fusing together takes place,
- 25 fig. 15 shows diagrammatically the movement pattern for the focal point of the radiation gun according to one embodiment of the invention,
- 30 fig. 16 shows a view from the side of a chamber provided with a transparent window,

- fig. 17 shows an arrangement for feeding and fixing a protective film for maintaining the transparency of the window,
- 5 fig. 18 shows a flow diagram for an operating scheme of the arrangement,
- fig. 19 shows a flow diagram for generating primary operating schemes,
- 10 fig. 20 shows a flow diagram for correction of said operating scheme,
- fig. 21 shows diagrammatically a procedure for generating a three-dimensional body,
- 15 fig. 22 shows diagrammatically a procedure comprising correction of operating schemes with the aid of information obtained from a camera which measures the temperature distribution over the surface of the powder bed,
- 20 fig. 23 shows a diagrammatic construction of a three-dimensional article, and
- fig. 24 shows a number of cross sections from figure 23.
- 25 MODE(S) FOR CARRYING OUT THE INVENTION

Figure 1 shows an arrangement for producing a three-dimensional product generally designated by 1. The arrangement comprises a work table 2 on which a three-dimensional product 3 is to be built up, one or more powder dispensers 4 and also means 28 which are arranged so as to distribute a thin layer of powder on the work table 2 for forming a powder bed 5, a radiation gun 6 for delivering energy to the powder bed, fusing together of parts of the

powder bed then taking place, means 7 for guiding the beam emitted by the radiation gun 6 over said work table for forming a cross section of said three-dimensional product by fusing together said powder, and a control computer 8 in which information about successive cross sections of the three-dimensional  
5 product is stored, which cross sections build up the three-dimensional product. In a work cycle, the work table will be lowered gradually in relation to the radiation gun after each powder layer applied. In order to make this movement possible, the work table is, in a preferred embodiment of the invention, arranged movably in the vertical direction, that is to say in the  
10 direction indicated by the arrow P. This means that the work table starts in a starting position 2' in which a first powder layer of the necessary thickness has been applied. So as not to damage the underlying work table and in order to provide this layer with sufficient quality, this layer is thicker than other layers applied, fusing through of this first layer then being avoided. The work  
15 table is subsequently lowered in connection with a new powder layer being distributed for forming a new cross section of the three-dimensional product. In one embodiment of the invention, the work table is to this end supported by a stand 9 which comprises at least one ball screw 10, provided with toothed  
11. A step motor or servomotor 12 provided with a gearwheel 13 sets the  
20 work table 2 to the desired vertical position. Other arrangements known to the expert for setting the working height of a work table can also be used. Adjusting screws, for example, can be used instead of racks.

The means 28 is arranged so as to interact with said powder dispensers for  
25 replenishment of material. Furthermore, the sweep of the means 28 over the working surface is driven in a known manner by a servomotor (not shown) which moves the means 28 along a guide rail 29 which runs along the powder bed.

When a new powder layer is applied, the thickness of the powder layer will be  
30 determined by how much the work table has been lowered in relation to the previous layer. This means that the layer thickness can be varied according to requirements. It is therefore possible, when a cross section has a great

change in shape between adjacent layers, to make thinner layers, a higher surface fineness then being achieved, and, when there is little or no change in shape, to make layers with maximum penetration thickness for the beam.

- 5 In a preferred embodiment of the invention, the radiation gun 6 consists of an electron gun, the means 7 for guiding the beam of the radiation gun consisting of deflecting coils 7". The deflecting coil 7" generates a magnetic field which guides the beam produced by the electron gun, it then being possible for fusion of the surface layer of the powder bed in the desired
- 10 location to be brought about. The radiation gun also comprises a high-voltage circuit 20 which is intended to provide the radiation gun in a known manner with an acceleration voltage for an emitter electrode 21 arranged in the radiation gun. The emitter electrode is in a known manner connected to a current source 22 which is used to heat the emitter electrode 21, electrons
- 15 then being released. The functioning and composition of the radiation gun are well-known to an expert in the field.

The deflecting coil is controlled by the control computer 8 according to an operating scheme laid out for each layer to be fused together, it then being possible to guide the beam according to a desired operating scheme.

- 20
- Also present is at least one focusing coil 7' which is arranged so as to focus the beam on the surface of the powder bed on the work table.

- 25
- Deflecting coils 7" and focusing coils 7' can be arranged according to a number of alternatives well known to the expert.

- 30
- The arrangement is enclosed in a casing 15 which encloses the radiation gun 6 and the powder bed 2. The casing 15 comprises a first chamber 23 which surrounds the powder bed and a second chamber 24 which surrounds the radiation gun 6. The first chamber 23 and the second chamber 24 communicate with one another via a passage 25 which allows emitted electrons, which have been accelerated in the high-voltage field in the second

chamber, to continue into the first chamber, subsequently to strike the powder bed on the work table 2.

In a preferred embodiment, the first chamber is connected to a vacuum pump  
5 26 which lowers the pressure in the first chamber 23 to a pressure of preferably roughly  $10^{-3} - 10^{-5}$  mbar. The second chamber 24 is preferably connected to a vacuum pump 27 which lowers the pressure in the second chamber 24 to a pressure of roughly  $10^{-4} - 10^{-6}$  mbar. In an alternative embodiment, both the first and second chambers can be connected to the  
10 same vacuum pump.

The control computer 8 is furthermore preferably connected to the radiation gun 6 for regulating the output of the radiation gun and also connected to the step motor 12 for adjusting the vertical position of the work table 2 between  
15 consecutive applications of powder layers, it then being possible to vary the individual thickness of the powder layers.

The control computer is also connected to said means 28 for powder distribution on the working surface. This means is arranged so as to sweep  
20 over the working surface, a layer of powder being distributed. The means 28 is driven by a servomotor (not shown) which is controlled by said control computer 8. The control computer controls the sweep along and ensures that powder is replenished as required. For this reason, load sensors can be arranged in the means 28, the control computer then being able to obtain  
25 information about the means being empty or having become stuck.

According to the invention, the control computer 8 is also arranged so as to calculate an energy balance for at least one selected area within each powder layer, it being determined in the calculation whether energy radiated into the  
30 part area from the surroundings of the part area is sufficient to maintain a defined working temperature of the part area.

According to a preferred embodiment of the invention, the control computer is also arranged so as to calculate an energy balance for at least the selected area to be fused together within each powder layer, it being determined in the calculation whether energy radiated into the selected area from the 5 surroundings of the selected area is sufficient to maintain a defined working temperature of the selected area.

The purpose of calculating the energy balance for the powder layers is to calculate the power required in order to keep the surface of the object at a 10 given temperature. The power is assumed to be constant over the entire surface.

How the energy balance calculation is performed in an embodiment of the invention where the calculation is performed for one layer at a time is 15 described below.

In order for it to be possible to calculate the power in real time, simplifications are necessary:

- 20 1. We imagine that the temperature is constant in the x and y directions and that it varies only in the z direction, in other words the entire surface has the same temperature.
2. The temperature in the z direction varies with  $jLt$ , where  $j$  is the layer number and  $Lt$  is the layer thickness.
- 25 3. The temperature distribution during fusion is assumed to be stationary.

The following parameters have an effect on the calculation:

- Various indexes:
- 30 1.  $i$  = index for the top layer
  2.  $j$  = layer index goes from 1 to  $i$

Object data:

$Lt$  = layer thickness to be fused [m]

$Lcont(j)$  = contour length for layer  $j$  [m]

$Apow(z) = Apow(j)$  = area facing the powder for layer  $j$  [m]

5  $A(z) = A(j)$  total surface area fused for layer  $j$  [ $m^2$ ]

Material properties:

$\lambda_{met}$  = thermal conductivity of the material [W/mK]

$\sigma_{met}$  = radiation constant for the metal surface [ $W/m^2K$ ]

10  $\sigma_{metpow}$  = radiation constant for metal surface covered with powder [ $W/m^2K$ ]

$\sigma_{pow}$  = radiation constant for the powder surface [ $W/m^2K$ ]

$\lambda_{pow}$  = thermal conductivity of the powder [W/mK]

$h_{pow}(z)$  = heat transfer coefficient from the object out to the powder [ $W/m^2K$ ]

15 Temperatures [K]:

$Tsur(i)$  = temperature of the surroundings affecting the surface when layer  $i$  is fused (can be measured on the heating shield)

$Tpow(z)$  = temperature in the powder

$T(z)$  = temperature in the object

20  $Tsurf(i) = T(iLt)$  = desired temperature on the surface of the object when layer  $i$  is fused. (Is set in AMA)

$Tbott(i)$  = temperature at the bottom of the object before layer  $i$  is started (Is measured just before raking or is calculated. See below.)

25 In order to determine how the temperature is distributed in the object, we solve the one-dimensional stationary thermal conductivity equation including a source term which takes account of heat losses out into the powder:

$$-\lambda_{met} \frac{\partial^2 T(z)}{\partial z^2} = \frac{h_{pow}(z)Apow(z)}{A(z)Lt} (Tpow(z) - T(z))$$

The boundary conditions on the surface and at the bottom are:

$$\begin{aligned}-\lambda_{met} \frac{\partial T(z)}{\partial z} \Big|_{z=iLt} &= \frac{(\sigma_{met} + \sigma_{pow})}{2} (T(iLt)^4 - T_{sur}(i)^4) - \frac{P_{in}}{A(iLt)} \\-\lambda_{met} \frac{\partial T(z)}{\partial z} \Big|_{z=0} &= h_{pow} (T_{bott}(i) - T(0))\end{aligned}$$

- 5 Where  $A$  and  $B$  are two constants.

Rewrite the formulas as differential formulas instead and let  $j$  be indexed for each layer.

$$\begin{aligned}-\lambda_{met} \frac{T(j+2) - 2T(j+1) + T(j)}{L_t^2} &= \frac{h_{pow}(j) A_{pow}(j)}{A(j)L_t} (T_{pow}(j) - T(j)) \\-\frac{\lambda_{met}}{L_t} (T(i) - T(i-1)) &= \frac{(\sigma_{met} + \sigma_{pow})}{2} (T(i)^4 - T_{sur}(i)^4) - \frac{P_{in}}{A(i)} \\-\frac{\lambda_{met}}{L_t} (T(2) - T(1)) &= h_{pow} (T_{bott}(i) - T(1))\end{aligned}$$

10

where

$$1 \leq j \leq i-2$$

- 15 The boundary condition on the surface actually provides us with nothing new as far as the temperature distribution in the object is concerned as the temperature of the surface is determined by  $T(i)$ . But it is required in order to determine  $P_{in}$  which is the power necessary in order to keep the temperature on the surface at  $T(i)$ .  $T(j)$  is now obtained from the following equation  
20 system:

$$\Delta(j) = -\frac{h_{pow}(j)A_{pow}(j)Lt}{A(j)\lambda_{met}}$$

$$T(j+2) - 2T(j+1) + T(j)(1 + \Delta(j)) = \Delta(j)Tp_{ow}(j)$$

$$T(1) = Tb_{ott}(i) \frac{h_{pow}(1)Lt/\lambda_{met}}{(1 + h_{pow}(1)Lt/\lambda_{met})} + T(2) \frac{1}{(1 + h_{pow}(1)Lt/\lambda_{met})}$$

Insert the expression pressure for  $T(1)$  and formulate the problem as a linear equation system:

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$$1. \quad T(3) - 2T(2) + T(2) \frac{(1 + \Delta(1))}{(1 + h_{pow}(1)Lt/\lambda_{met})} = \Delta(1)Tp_{ow}(1) - Tb_{ott}(i) \frac{h_{pow}(1)Lt/\lambda_{met}}{(1 + h_{pow}(1)Lt/\lambda_{met})} (1 + \Delta(1))$$

$$2. \quad T(4) - 2T(3) + T(2)(1 + \Delta(2)) = \Delta(2)Tp_{ow}(2)$$

$$i-2. \quad -2T(i-1) + T(i-2)(1 + \Delta(i-2)) = \Delta(i-2)Tp_{ow}(i-2) - T(i)$$

In matrix form this becomes:

$$10 \quad Ax = b$$

where then:

$$A_{jk} = \delta(j+1-k) - 2\delta(j-k) + \delta(j-1-k)(1 + \Delta(j)) + \delta(1-k)\delta(1-j) \frac{(1 + \Delta(1))}{(1 + h_{pow}(1)Lt/\lambda_{met})}$$

$$x_1 = T(2), \dots, x_{i-2} = T(i-1)$$

$$b_j = \Delta(j)Tp_{ow}(j) - \delta(j-1)Tb_{ott}(i) \frac{h_{pow}(1)Lt/\lambda_{met}}{(1 + h_{pow}(1)Lt/\lambda_{met})} (1 + \Delta(1)) - \delta(j-i+2)T(i)$$

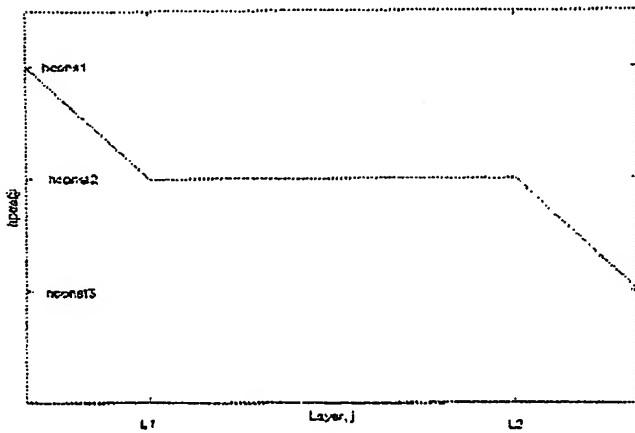
In order for it to be possible to solve the equations, it is necessary that the temperature of the powder,  $T_{pow}(j)$ , and the heat transfer coefficient,  $h_{pow}(j)$ , are known. In the program,  $T_{pow}(z)$  is set to:

$$T_{pow}(j) = AT(j)_{i-1} + BT_{sur}(i-1)$$

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$i-1$  means that the temperature for the previous layer is used in order to determine  $T_{pow}(j)$ .

- 10 The function used for  $h_{pow}(j)$  looks like this:



- 15 The values  $L1$  and  $L2$  have been assumed to be area-independent while  $h_{const1}$ ,  $h_{const2}$  and  $h_{const3}$  are assumed to depend on  $A(j)$ . All the constants in the expressions for both  $T_{pow}$  and  $h_{pow}$  have been produced by adapting the 1D model above to 3D FEM calculations on objects with simple geometries.
- 20 Included in the expression for the source term is  $Apow(j)$  which is actually the total area facing the powder for each layer. In the case of large area

transitions, this value may be very great, which means that the value of the source term jumps. Such discrete jumps make the solution unstable. In order to prevent this, according to a preferred embodiment,  $A_{pow}(j)$  is set to  $L_{cont}(j)*Lt$ . Power losses which arise owing to an area transition are instead

5 added afterwards. The size of the power loss depends on how large over the respective underlying area the area transition is and how far below the top layer it is located. The values for different area transitions and different depths have been produced by 3D FEM simulations. For an arbitrary area transition, the additional power is obtained by interpolation.

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Before the power is calculated, the program reads the various values for  $L_{cont}(j)*Lt$  and  $A(j)$  for each layer. With the aid of a script file, these can be influenced in various ways. In this way, it is possible to control the power for each layer. How the various geometry parameters are influenced emerges  
15 from the description of how the script file functions.

When the equation system above is solved, the total power required in order to keep the surface at  $T_{surf}(i)$  is obtained from the boundary condition for the surface:

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$$P_{in} = A(i) \left( \frac{\lambda_{met}}{Lt} (T(i) - T(i-1)) + \frac{(\sigma_{met} + \sigma_{pow})}{2} (T(i)^4 - T_{sur}(i)^4) \right)$$

When a layer is fused, use is made of different current and speed over the surface. In order for it to be possible to calculate the different currents  
25 required, the mean value of all powers used is set equal to  $P_{in}$ .

If a layer is to be fused using  $n_i$  different currents, then:

$$P_{in} = \alpha U \frac{\sum_{k=1}^{k=n_i} I_{ik} t_{ik}}{T_{tot}}$$

$$t_{ik} = \frac{l_{ik}}{v_{ik}}$$

$$T_{tot} = \sum_{k=1}^{k=n_i} t_{ik}$$

Where  $t_{ik}$  is the fusion time for each current  $I_{ik}$

$I_{ik}$  is the fusion length

5  $v_{ik}$  is the fusion speed

$T_{tot}$  is the total fusion time for the layer  $i$

$U$  is the acceleration voltage.

In order for it to be possible to calculate the currents, the speeds must  
 10 therefore be known. These are obtained from what are known as speed  
 functions which indicate the relationship between current and speed. As these  
 functions are not analytical, an iterative procedure must be used in order to  
 determine all the currents and speeds. In the calculation program, each  
 starting value of  $I_{ik}$  is guessed. The various speeds are then obtained. The  
 15 values of the currents are then increased until the mean value of the power  
 just exceeds the calculated value of  $P_{in}$ .

Assume now that we want to fuse the various part areas at such a speed and  
 20 current that the energy which is delivered to the material is less than that  
 required in order to keep the surface at  $T_{surf}(i)$ . The surface must then be  
 heated. The number of times required in order to heat the surface is obtained  
 by adding a heating term in the expression for the mean value of the power:

$$P_i^{\text{heat}} = \frac{n \alpha U l_i^{\text{heat}} I_i^{\text{heat}}}{v_i^{\text{heat}}}$$

and adding the heating time in the expression for the time  $T_{\text{tot}}$ :

$$t_i^{\text{heat}} = \frac{n l_i^{\text{heat}}}{v_i^{\text{heat}}}$$

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where  $n$  indicates how many times the surface has to be heated.

- The calculation routine shown above can be used for the entire powder layer.
- 10 In an alternative embodiment, calculation can be carried out for various part areas of the powder layer. The equations indicated above can be used in this case as well. However, different boundary conditions are obtained for the inner edges which lie close to a fused body.
  - 15 Figure 2 shows a side view of a fused-together body 30 which is built up by fusing together part areas 31-34 in consecutive powder layers i-1, i-2, i-3, i-4. A real body manufactured according to the invention can of course have many more layers than indicated in this example.
  - 20 A top powder layer i is distributed on the body. Located within this top powder layer is a selected area 35. The selected area 35 consists of the area which, according to an operating scheme, is to be fused together. The selected area 35 within the layer i is delimited by an outer edge 36. It is of course conceivable for a selected area to comprise both outer and inner edges.
  - 25 Within the selected area 35, a part area 37 is shown, for which an energy balance is to be calculated. The part area 37 can consist of a part of the

selected area as shown in figure 2 or alternatively can consist of the entire selected area. The selected area 35 is preferably divided into a plurality of smaller part areas as shown in figure 6.

- 5      The energy balance is calculated in principle according to  $E^{in}(i) = E^{out}(i) + E^{heat}(i)$ , where  $E^{in}(i)$  represents energy fed into the part area,  $E^{out}(i)$  represents energy losses through dissipation and radiation from the part area, and  $E^{heat}(i)$  represents stored in the part area. The energy fed in consists of on the one hand energy  $E^{in(c)}$  which has been radiated in or has flowed in via
- 10     thermal conduction from the surroundings of the part area 35 for which the energy balance is calculated and on the other hand of energy  $E^{in(s)}$  which has been radiated in from the radiation gun 6. If the energy balance is calculated before energy has been supplied to the part area 35,  $E^{in(s)}$  therefore = 0. According to a preferred embodiment of the invention, at least a first energy
- 15     balance calculation is performed for the part area 35 before energy has been supplied via the radiation gun 6.

- 20     Figure 3 shows diagrammatically a model on which the calculation of the energy balance for the part area  $\Delta_1$  is based. In this case, the part area  $\Delta_1$  corresponds to a part of the selected area of the powder layer i. In this case, the equation for calculation of the energy balance has the appearance  $E^{in}(\Delta_1) = E^{out}(\Delta_1) + E^{heat}(\Delta_1)$ , where  $E^{in}(\Delta_1)$  represents energy fed into the part area,  $E^{out}(\Delta_1)$  represents energy losses through dissipation and radiation from the part area  $\Delta_1$  and  $E^{heat}(\Delta_1)$  represents stored in the part area  $\Delta_1$ . The energy fed in consists of on the one hand energy  $E^{in(c)}(\Delta_1)$  which has been radiated in or has flowed in via thermal conduction from the surroundings of the part area  $\Delta_1$  and on the other hand of energy  $E^{in(s)}(\Delta_1)$  which has been radiated in from the radiation gun 6.
- 25     Figure 4 shows diagrammatically a model on which the calculation of the energy balance for a second part area  $\Delta_2$  within the selected area 35 in the

layer i is based. In this case, the part area  $\Delta_2$  corresponds to a part of the selected area 35 of the powder layer i which has not yet been fused together and which is adjacent to a first part area  $\Delta_1$  within the powder layer i, where radiation or thermal conduction takes place from said first to said second part

5 area. In this case, the equation for calculation of the energy balance has the appearance  $E^{in}(\Delta_2) = E^{out}(\Delta_2) + E^{heat}(\Delta_2)$ , where  $E^{in}(\Delta_2)$  represents energy fed into the part area,  $E^{out}(\Delta_2)$  represents energy losses through dissipation and radiation from the part area  $\Delta_1$  and  $E^{heat}(\Delta_2)$  represents stored in the part area  $\Delta_2$ . The energy fed in consists of on the one hand energy  $E^{in(c)}(\Delta_2)$

10 which has been radiated in or has flowed in via thermal conduction from the surroundings of the part area  $\Delta_1$  and on the other hand of energy  $E^{in(s)}(\Delta_2)$  which has been radiated in from the radiation gun 6. The energy  $E^{in(c)}(\Delta_2)$  supplied via thermal conduction comprises the component  $E^{in(s)}(\Delta_2, i-1)$  which corresponds to energy supplied from the previous layer and also  $E^{out}$

15 ( $\Delta_1, \Delta_2$ ) which corresponds to energy which has been dissipated or radiated from the first part area  $\Delta_1$  and supplied to the second part area  $\Delta_2$ .

Figure 5 shows the movement of two focal points 201, 202 within the selected area 203. The selected area 203 therefore comprises two fusion zones 204, 205 which propagate simultaneously when formation of a cross section of the three-dimensional body takes place. The propagation of the fusion zones 204, 205 is shown diagrammatically by the movement of the front lines 206, 207 of the fusion zones. The figure shows the state of the selected area on six occasions. On a first occasion, the radiation gun 6 delivers energy to the focal point I, the front line I being formed. On a subsequent occasion, the focal point of the radiation gun has been moved to the focal point II, the front line II being formed. In the next stage, the focal point I is moved to the focal point III, that is to say by an increment  $\Delta$ . During the same time, the front line I has been moved by an increment  $\delta$  to the front line III. In a preferred embodiment 25 of the invention, the increment  $\delta$  is the same as the increment  $\Delta$ . In the next step, the focal point II is moved to the focal point IV, the front line II being

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moved to the front line IV. In another subsequent stage, the focal point III is moved to the focal point V, and the front line III to the front line V. Shown finally is how the focal point IV is moved to the focal point VI while the front line IV moves to the front line VI.

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As shown in figure 5, the movement between three focal points does not have to be rectilinear but can be controlled so as to adapt so that the front lines fill the selected area in the most rapid way possible.

- 10    In order to adapt the speed of the propagation of the focal point to the propagation speed of the fusion zones, that is to say the fusion zone wave propagation speed, the wave propagation speed is, according to one embodiment of the invention, estimated from information provided by means for sensing the temperature distribution of a surface layer located in the
- 15    powder bed. This means can consist of a heat camera which can measure the position of the front line of the fusion zone on two consecutive occasions, in which way the speed can be calculated.

- 20    In an alternative embodiment, the wave propagation speed is estimated by calculating an energy balance for an area comprising said focal points, said wave propagation speed being obtained from a model of a thermal conductivity equation set up for said area.

- 25    According to a first embodiment of the invention, the operating scheme stored in the control computer is arranged in such a way that said number of focal points are spread out over the entire selected area corresponding to the entire cross section of the three-dimensional body under formation. According to an alternative embodiment, several focal points are used within the part areas as described below.

30

Figure 6 shows a selected area 35 which is divided into a plurality of smaller part areas. According to a preferred embodiment of the invention, the surface

within each powder layer is divided into a set of separate areas 38-53 which each comprise some part of the selected area 35, an energy balance being calculated for each of said set of separate areas 38-53. The selected area is delimited by an outer edge 72. The selected area can of course also comprise  
5 inner edges.

According to another preferred embodiment of the invention, said set of separate areas 38-53 comprises a first group of areas 54-58 which lie entirely within the edge 72 of said selected area and a second group of areas 38-53  
10 of which the edges coincide in part with the edge of said selected area. Where appropriate, each of the areas 38-53 included in said second group is divided into a second set of smaller areas 38a-38d, 53a-53d.

Figure 7 shows diagrammatically the operating scheme according to a  
15 preferred embodiment of the invention where each area 80-91 within said first and second group of areas consists of an inner area I and an edge R where the inner area I of a set of adjacent areas within said first or second group of areas is fused together in a first process step, after which said edges R between said areas are fused together. By means of this procedure, the  
20 occurrence of bending stresses in the three-dimensional body after cooling is reduced.

According to a preferred embodiment of the invention, the operating scheme is designed in such a way that the area to be fused together, that is to say the  
25 selected area, is divided into one or more inner areas I, each having an edge R. Figure 8 shows diagrammatically an area 35 to be fused together. The area comprises an inner area I which is delimited by an edge R. According to the invention, the inner area I is fused together using a movement pattern for the focal point of the beam of the radiation gun which comprises a main  
30 movement direction and an interference term which is added to said main movement direction and has a component in a direction at right angles to the main movement direction. The interference term changes direction and has a

time mean value corresponding to zero drift from the main movement direction. Figure 9 shows three different examples of different appearances of the interference term which give rise to a movement in the form of a triangular wave, a sinusoidal curve and a square wave.

5

Figure 10 shows diagrammatically how the heat distribution appears in a body where the focal point with the diameter D of a radiation gun has heated the body. The temperature distribution around the focal point has the shape of a Gaussian bell. The temperature distribution around a focal point without 10 interference term is shown by the curve marked ( $\alpha$ ). By means of the interference term, the trace treated in the course of propagation of the beam along the main movement direction is widened. A widened trace is shown by the curve indicated by ( $\beta$ ). The widened trace also has a temperature distribution with a lower maximum value. This reduces the risk of the 15 appearance of overheating with the formation of irregularities as a consequence.

The interference term is preferably of such a nature that a fusion zone is formed which has a width essentially corresponding to twice the amplitude of 20 the component of the interference term in a direction at right angles to the main movement direction. The average speed of the absolute value of the movement of the focal point in the direction of the interference term is preferably to exceed the speed of the heat propagation in the material. The speed in the main movement direction preferably corresponds to the speed of 25 the heat propagation in the material. The amplitude and the frequency of the interference term are preferably to be adapted in such a way that the focal point is able to move from its starting position where the interference term has the value zero, pass through the minimum and the maximum value of the interference term and return to its position in the time it takes the wave front 30 of the heat propagation to move from the first zero position to the second zero position. This is shown diagrammatically in figure 5. Figure 11 shows how the focal point moves along the curve 50 from a first position 51, past a maximum

52 of the interference term, a minimum 53 of the interference term and then takes up a second position 54 with a zero value of the interference term. During this time, the wave front of the heat propagation has been propagated from the first position 51 to the second position. If the average speed of the  
5 interference term is too low, a curved fused trace which runs within the path defined by the end points of the interference term is formed instead of a wide trace.

According to a preferred embodiment, the interference term also has a  
10 component in a direction parallel to the main movement direction. The interference term is in this case two-dimensional. Examples of interference terms with a two-dimensional direction are given in figure 12.

The edge R is preferably fused together in the course of a mainly rectilinear  
15 movement of the beam of the radiation gun.

The purpose of operating with a movement pattern for the focal point of the beam of the radiation gun which comprises a main movement direction and an interference term added to said main movement direction which has a  
20 component in a direction at right angles to the main movement direction is that, with a wider trace, it is possible to move the fusion zone more slowly but still fuse at a relatively high speed compared with conventional operation. Slow movement of the fusion zone produces less vaporization and a reduced incidence of fused material boiling and splashing. The purpose of the edge  
25 being fused together using a continuous mainly rectilinear movement is that this produces a smooth surface structure for the finished product.

An analysis of the movement pattern for the beam of the radiation gun in the case of a preferred embodiment of the invention with a two-dimensional  
30 interference term, which gives rise to a helix-like movement pattern of the focal point, follows below.

The position of a focal point which rotates about the x axis and moves along the same axis at the speed  $V_x$  can be obtained from:

$$\vec{r}(t) = (V_x t + A_x \cos(\omega t))\vec{x} + A_y \sin(\omega t)\vec{y} \quad \text{Equ. 1}$$

5

where  $A_x$  and  $A_y$  are the amplitudes in the x and y direction respectively.  
A typical "spinning curve" can appear like that shown in figure 13.

The pattern shown in figure 7 is obtained if  $\omega$  is set to:

10

$$\omega = \frac{2\pi V_x}{A_x} \quad \text{Equ. 2}$$

The speed of the focal point is given by:

$$\frac{d\vec{r}(t)}{dt} = (V_x - A_x \omega \sin(\omega t))\vec{x} + A_y \omega \cos(\omega t)\vec{y} \quad \text{Equ. 3}$$

15

Its absolute speed is therefore:

20

$$\left| \frac{d\vec{r}(t)}{dt} \right| = \sqrt{(V_x - A_x \omega \sin(\omega t))^2 + (A_y \omega \cos(\omega t))^2} \quad \text{Equ. 4}$$

If the focal point moves according to the formulas above, its speed will vary and either be at a maximum underneath the x axis and a minimum above or vice versa depending on the direction of rotation. In order to obtain a focal

point which moves at constant speed along the spinning curve in figure 1, its average speed is first calculated:

$$V_{\text{average}} = \frac{\int_0^T \left| \frac{dr(t)}{dt} \right| dt}{T} \quad \text{Equ. 5}$$

5

Where:

$$T = \frac{2\pi}{\omega}$$

- 10  $V_{\text{average}}$  is the speed at which the focal point is to move. At the time  $t$ , the focal point has moved the distance:

$$s = t * V_{\text{average}}$$

- 15 This distance must be equal to the spinning curve length at the time  $t'$ . Therefore:

$$s = t * V_{\text{average}} = \int_0^{t'} \left| \frac{dr(t)}{dt} \right| dt \quad \text{Equ. 6}$$

- 20 Solving Equ. 6 for  $0 < t < T$  gives  $t'$  as a function of  $t$ .  $t'$  is then used in Equ. 1 which gives the position of the spot as a function of the time  $t$ .

A number of simulations using different speeds and Ay has shown that the fusion zone 0.1-0.15 mm below the surface has an approximate width of 1.8Ay. The hop between two spin lines should then be:

- 5    *Hop spin=1.8Ay-0.3*

The distance to the start from an edge is approximately:

*Starting hop=0.8Ay-0.15*

10

Figure 14 shows a continuous wide fused edge which propagates in the direction x marked by an arrow in the diagram. The focal points with a diameter D are marked in the diagram. The overlapping pattern ensures that fusing together takes place within an area outside the focal point. Such an 15 area is illustrated and marked by the symbol  $\delta$ . Together, these areas form an overall area which propagates in the direction marked by the arrow x.

According to a further preferred embodiment of the invention, the edges R between the inner areas I are fused together in the course of a mainly 20 rectilinear movement of the beam of the radiation gun.

According to fig. 15, the arrangement also comprises, according to a preferred embodiment of the invention, means 14 for sensing surface properties of a surface layer located in the powder bed. This means 14 for 25 sensing the temperature distribution of a surface layer located in a powder bed 5 preferably consists of a camera. In a preferred embodiment of the invention, the camera is used on the one hand to measure the temperature distribution on the surface layer and on the other hand to measure the occurrence of surface irregularities by means of the shadow formation to 30 which surface irregularities give rise. On the one hand, information about the temperature distribution is used to bring about as uniform a temperature distribution as possible over those parts of the surface layer which are to be

- fused and, on the other hand, information can be used in order to check for any dimensional deviations between generated three-dimensional product and original design as the temperature distribution reflects the shape of the product.
- In a preferred embodiment of the invention, the video camera is
- 5 mounted on the outside of the casing 15 which encloses the powder bed 5 and the radiation gun 6. In order to make temperature measurement possible, the casing is provided with a transparent window 16. The powder bed 5 is visible for the camera through this window.
- 10 In a preferred embodiment of the invention, which is shown in figure 16, the window 16 is covered by a protective film 17. The protective film is fed from a feed-out unit 18 to a collecting unit 19, the film being gradually replaced, which means that the transparency can be maintained. The protective film is necessary as coatings form as a consequence of the fusion process.
- 15
- Figure 17 shows diagrammatically a method of producing three-dimensional bodies according to the invention. The three-dimensional body is formed by successive fusing together of selected areas of a powder bed, which parts correspond to successive cross sections of the three-dimensional body.
- 20
- In a first method step 100, application of a powder layer to a work table takes place. Application is effected by the means 28 mentioned above distributing a thin layer of powder on the work table 2.
- In a second method step 110, energy is supplied from a radiation gun 6,
- 25 according to an operating scheme determined for the powder layer, to a selected area within the powder layer, fusing together of the area of the powder layer selected according to said operating scheme then taking place to form a cross section of said three-dimensional body. A three-dimensional body is formed by successive fusing together of successively formed cross
- 30 sections from successively applied powder layers.

According to the invention, said operating scheme is designed in such a way that the selected area has two or more fusion zones which propagate simultaneously through the selected area when formation of a cross section of the three-dimensional body takes place.

5

The two or more fusion zones are brought about by the radiation gun supplying energy to two or more geometrically separate focal points while time sampling takes place.

- 10 The operating scheme is preferably designed in such a way that the focal points of the radiation gun at the fusion points propagate at a speed which corresponds to the wave propagation speed of the fusion zone.

According to one embodiment of the invention, the wave propagation speed  
15 is estimated by measuring the wave propagation speed of the fusion zone from information provided by means for sensing the temperature distribution of a surface layer located in the powder bed. By reading the temperature distribution over the surface of the body, this means can record how the heat spreads from the fusion points and outward. The speed at which the heat  
20 spreads corresponds to the wave propagation speed.

According to a preferred embodiment, an energy balance is calculated in a third method step 120 for at least one part area within each powder layer, it being determined in the calculation whether energy radiated into the part area  
25 from the surroundings of the part area is sufficient to maintain a defined working temperature of the part area. Calculation is performed according to the models indicated above.

Figure 18 shows diagrammatically the procedure for generating primary  
30 operating schemes. In a first step 40, a 3D model is generated, in a CAD program for example, of the product to be manufactured, or alternatively a ready-generated 3D model of the product to be manufactured is input into the

control computer 8. Then, in a second step 41, a matrix containing information about the appearance of cross sections of the product is generated. Figure 23 shows a model of a hammer with examples of associated cross sections 31-33. These cross sections are also shown in 5 fig. 24a-24c. The cross sections are distributed with a density corresponding to the thickness of the various layers to be fused together in order to form the finished product. The thickness can advantageously be varied between the various layers. It is inter alia advantageous to make the layers thinner in areas where there is great variation in the appearance of the cross sections 10 between adjacent layers. When the cross sections are generated, a matrix containing information about the appearance of all the cross sections which together make up the three-dimensional product is therefore created.

Once the cross sections have been generated, a primary operating scheme is 15 generated for each cross section in a third step 42. Generation of primary operating schemes is based on shape recognition of the parts which make up a cross section on the one hand and knowledge of how the operating scheme affects the cooling temperature of local parts of a cross section on the other hand. The aim is to create an operating scheme which allows the cooling 20 temperature to be as uniform as possible in the parts which are fused together before the next layer is applied at the same time as the cooling temperature is to be kept within a desired range in order to reduce the risk of shrinkage stresses appearing in the product and to reduce the magnitude of shrinkage stresses which have arisen in the product, with deformation of the 25 product as a consequence.

In the first place, a primary operating scheme is generated on the basis of the shape of separate component parts of the cross section.

30 In a preferred embodiment of the invention, primary operating schemes are therefore laid out on the basis of experience of which operating schemes provide a good temperature distribution of the cooling temperature of the

cross section, it then being possible for the risk of shrinkage stresses in the product with deformation of the product as a consequence to be reduced. To this end, a set of operating schemes for areas of different shapes is stored in a memory. In a preferred embodiment, this memory is updated as results of

- 5 corrections of the operating scheme are evaluated, a self-learning system being obtained.

In an alternative embodiment of the invention, ready-finished cross sections, which have been generated by a stand-alone computer, are input into a  
10 memory in the control computer, where said primary operating schemes are generated. In this case, information is provided directly to the third step 42 via an external source 40a.

Figure 19 shows diagrammatically a procedure for generating a three-dimensional body, which uses the method described in connection with figure  
15 17. In a first method step 130, parameters on which an energy balance calculation is based are determined. In a second method step 140, calculation of the energy balance for at least one part area of the selected area 35 takes place. Calculation is performed according to the method  
20 illustrated previously.

In a third method step 150, the operating scheme is updated depending on the calculated energy balance. If the result of the energy balance is that sufficient heat energy is stored in the part area to maintain a desired working  
25 temperature, no extra energy supply takes place. According to one embodiment of the invention, if the result of the energy balance is that sufficient heat energy to maintain a desired working temperature is not stored in the part area, an extra energy supply takes place in the form of preheating of the part area before fusing together takes place. This preheating can be  
30 effected by the radiation gun being swept very rapidly over the area or the radiation gun sweeping over the area with lower power than normal, or

alternatively a combination of both of these. The preheating takes place in a fourth method step 160.

In a fifth method step 170, fusing together is effected by the radiation gun sweeping over the part area.

5

Figure 20 shows diagrammatically an embodiment of the invention which, where appropriate, utilizes the methods described above for generating and correcting the operating schemes. In a first method step 180, one or more of the inner areas I of the selected area are identified. In a second method step 10 190, the edge or edges R which are associated with said inner areas and each surround said inner areas are identified. In a third method step 200, said inner areas I are fused together in the course of a partly overlapping circular movement of the beam emitted by the radiation gun. During a fourth method step 210, said edges are fused together in the course of a rectilinear 15 movement of the beam.

According to one embodiment of the invention, the operating scheme is arranged so as consecutively to fuse together the powder within one area at a time within said inner areas.

20

According to a preferred embodiment of the invention, the control computer is arranged so as to divide the surface within each powder layer into a set of separate areas, said set of separate areas comprising a first group of areas which lie entirely within said selected area and a second group of areas which 25 lie partly within said selected area, where each area within said first and second group of areas consists of an inner area with an edge. The control computer is also arranged so as to ensure that said inner areas of a set of areas within said first group of areas are fused together in the course of a partly overlapping circular movement of the beam of the radiation gun.

30

In one embodiment of the invention, the fusing together of the inner areas in said second group of areas takes place with a partly overlapping circular

movement. In an alternative embodiment of the invention, the fusing together within the inner areas in said second group of areas takes place with a mainly rectilinear movement.

- 5 According to a preferred embodiment of the invention, in the case of the embodiments described above relating to division of the selected surface into smaller part surfaces, the calculation of energy balance described above is used in order to control the operating scheme with regard to calibration of the power of the beam and supply of energy for heating the powder bed before
- 10 final fusing together takes place.

Figure 21 shows diagrammatically a procedure where the beam from the radiation gun is guided over the powder bed in order to generate a cross section of a product. In a first step 50, guidance of the beam over the powder bed according to the primary operating scheme defined in step 42 is started. In the next step 51, the temperature distribution on the surface layer of the powder bed is measured by the camera. From the measured temperature distribution, a temperature distribution matrix,  $T_{ij\text{-measured}}$ , is then generated, in which the temperature of small part areas of the surface layer of the powder bed is stored. When the matrix is generated, each temperature value  $T_{ij\text{-measured}}$  in the matrix is compared with a desired value in a desired value matrix  $T_{ij\text{-desired value}}$ . The surface layer of the powder bed can be roughly divided into three categories. Firstly, areas where fusing together takes place by treatment by the radiation gun. In these areas, the maximum fusion temperature  $T_{ij\text{-max}}$  is of interest. Secondly, areas which have already been fused together and are thus cooling. In these areas, a minimum permitted cooling temperature  $T_{ij\text{-cooling-min}}$  is of interest because too cold a cooling temperature gives rise to stresses and thus deformations of the surface layer. Thirdly, areas which have not been treated by the radiation gun. In these areas, the bed temperature  $T_{ij\text{-bed}}$  is of interest. It is also possible for the temperature to be compared only in treated areas,  $T_{ij\text{-bed}}$  not then being stored and/or checked.

In a third step 52, it is investigated whether  $T_{ij\text{-measured}}$  deviates from the desired value  $T_{ij\text{-desired value}}$  and whether the deviation is greater than permitted limit values. Limit values  $\Delta T_{ij\text{-max}}$ ,  $\Delta T_{ij\text{-cooling}}$  and  $\Delta T_{ij\text{-bed}}$  associated with the

5 three different categories are stored in the control computer 8. It is also possible for the bed temperature not to be checked. In this case, the associated limit value is not stored. If the deviation between  $T_{ij\text{-measured}}$  and  $T_{ij\text{-desired value}}$  does not exceed this limit value, it is investigated in a fourth step 53 whether the surface layer is fully treated. If this is not the case, operation

10 according to the current operating scheme continues, method steps 50-53 mentioned above being run through once again.

If the deviation between  $T_{ij\text{-measured}}$  and  $T_{ij\text{-desired value}}$  exceeds one of said limit values, correction of the operating scheme 42 takes place in a fifth step. In a

15 preferred embodiment, said correction is carried out according to the system shown in figure 17.

In a preferred embodiment of the invention, a new powder layer is distributed only after completion of each layer, the product being built up by successive

20 fusing together of powder layers until the product is finished. In this case, after a sixth step 55, a new layer is started, if the product as a whole is not finished, when it has been established in the fourth step 53 that the operating scheme for a layer has been completed.

25 In a preferred embodiment, which is shown in figure 22, the correction of the operating scheme comprises the following method steps:  
in a first step 56,  $T_{ij\text{-max}}$  is compared with  $T_{ij\text{-max-desired value}}$ . If  $T_{ij\text{-max}}$  deviates from  $T_{ij\text{-max-desired value}}$  exceeding  $\Delta T_{ij\text{-max}}$ , the energy supply to the powder layer is calibrated in a step 56a by either changing the power of the beam or

30 changing the sweep speed of the beam.

In a second step 58,  $T_{ij\text{-cooling}}$  is compared with  $T_{ij\text{-cooling-desired}}$  value. If  $T_{ij\text{-cooling}}$  deviates from  $T_{ij\text{-cooling-desired}}$  value exceeding  $\Delta T_{ij\text{-cooling}}$ , the operating scheme of the beam is changed in a step 58a. There are many ways of changing the operating scheme of a beam. One way of changing the operating scheme is

- 5 to allow the beam to reheat areas before they have cooled too much. The radiation gun can then sweep over areas already fused together with a lower energy intensity and/or at a higher sweep speed.

In a third step 60, it is investigated whether  $T_{ij\text{-bed}}$  deviates from  $T_{ij\text{-bed-desired}}$  value. If the deviation is greater than  $\Delta T_{ij\text{-bed}}$ , the temperature of the bed can, in one embodiment of the invention, be corrected in a step 60a, for example by the beam being made to sweep over the bed to supply energy. It is also possible to connect separate bed-heating equipment to the arrangement.

- 10
- 15 It is also possible for a size check of the article being manufactured to be carried out by the heat camera installed in the arrangement. As described above, the bed and the parts which have been fused together are measured. The measured heat distribution reflects fully the shape of the object in a section of the three-dimensional body to be created. A check of the dimensions of the article can in this way be carried out in a fourth step 62, and feedback of X-Y deflection of the beam of the radiation gun is thus possible. In a preferred embodiment of the invention, this check of the deviation between dimensions of the cross section is carried out in a step 62a and, if the deviation is greater than permitted, the X-Y deflection of the
- 20
- 25 radiation gun is corrected.

Moreover, input signals from the camera can be used for identifying the occurrence of surface irregularities, for example in the form of a welding spark. When the coordinates of a surface irregularity have been identified, the operating scheme can be updated so that the radiation gun is ordered to the identified coordinate in order to melt down the surface irregularity.

The invention is not limited to the illustrative embodiment described above; the radiation gun can consist of, for example, a laser, in which case the deflection means consist of guidable mirrors and/or lenses.

- 5    The invention can furthermore be used in an arrangement for producing a three-dimensional product by energy transfer from an energy source to a product raw material, which arrangement comprises a work table on which said three-dimensional product is to be built up, a dispenser which is arranged so as to distribute a thin layer of product raw material on the work
- 10    table for forming a product bed, a means for delivering energy to selected areas of the surface of the product bed, a phase transition of the product raw material being allowed for forming a solid cross section within said area, and a control computer which manages a memory in which information about successive cross sections of the three-dimensional product is stored, which
- 15    cross sections build up the three-dimensional product, where the control computer is intended to control said means for delivering energy so that energy is supplied to said selected areas, said three-dimensional product being formed by successive joining together of successively formed cross sections from product raw material applied successively by the dispenser.

20

- In this case, the embodiment is not limited to fusing together powder by a radiation gun irradiating the surface of a powder bed. The product raw material can consist of any material which forms a solid body after a phase transition, for example solidification after fusion or hardening. The energy-delivering means can consist of an electron gun or a laser guided over the working surface or alternatively of an energy-delivering means which can project a cross section directly onto the product bed.

- 25    The embodiment described above can moreover be provided with all the features described in relation to the embodiments described previously.